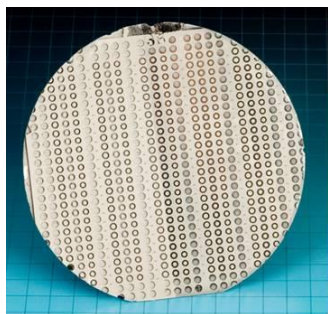


# Pressure, Temperature, and Recession Measurements in Extreme Environments

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**4" SiC wafer with  
pressure sensors**



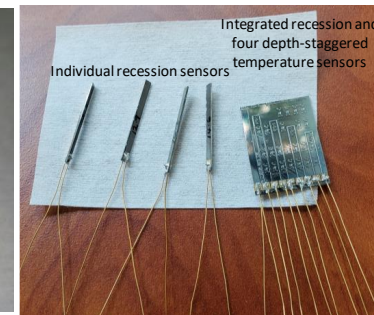
**Swagelok assembled SiC  
Sensors**



**Through bore assembled  
SiC Sensors**



**NPT assembled SiC Sensors**



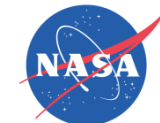
**Recession sensors**



# Outline

- **Application Drivers**
- **Why SiC for High Temperature Sensing?**
- **Technical/Technological Challenges**
- **SiC Pressure Sensor Characterization**
- **Field Validation**
- **Recession Sensor Characterization**
- **Conclusion**

# Application Drivers



**Turbine Engines**

- Thermoacoustic instabilities
- Active combustion control
- Exhaust noise emission



**X-43A Scramjet with supersonic Combustion ramjet engine**

- Combustion instabilities
- Mode Transition
- Unstart



## Venus

- 460 C
- Aggressive chemistry
- Atmospheric pressure (93 Earth atm.)

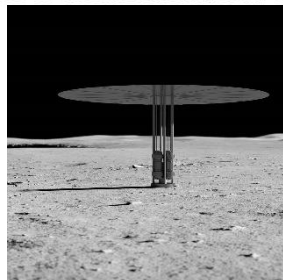


Recession sensors for Thermal Protection Systems

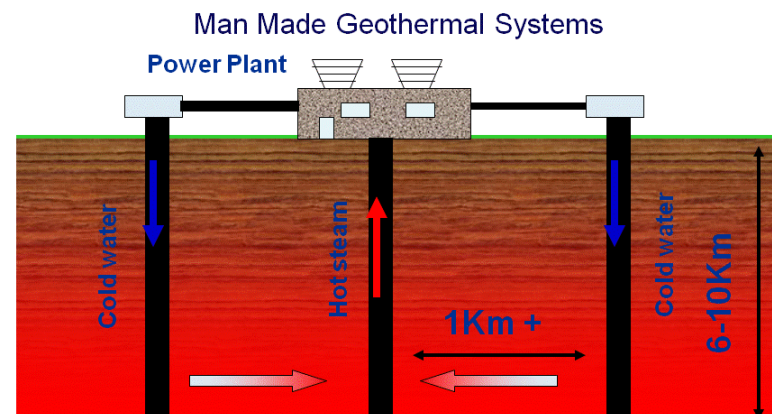


Pressure pulse/shock quantification

**Pulse detonation engine**



Instrumentation for Advanced Micro Nuclear Reactors

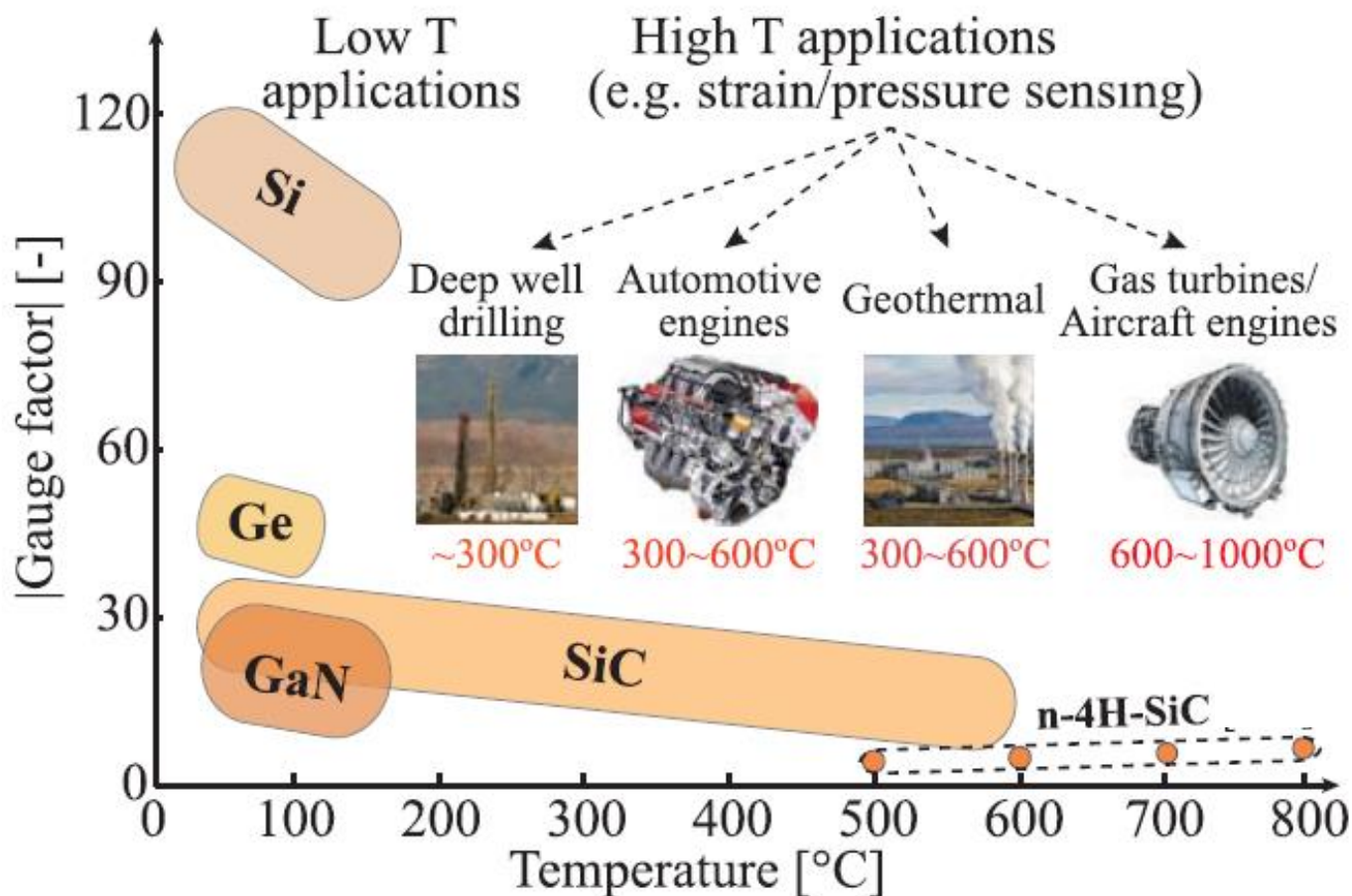


- The Earth is hot. At depths of 6 to 8Km temperatures exceed 200C
- Geothermal tools require electronics to operate from 140 – 325C
- The heat is mined by water flow from injectors to the producer via hydraulic enhanced fractures in the rock.

# Why SiC for High Temperature Sensing?

Properties	SiC	Si	Diamond	GaN
Energy gap (eV)	2.3 (3C-SiC) to 3.4 (2H-SiC)	1.12	5.5	3.4
Breakdown voltage (V/cm)	$4 \times 10^6$	$3 \text{ to } 6 \times 10^5$	$10 \times 10^6$	$3 \times 10^6$
Electron mobility ( $\text{cm}^2/\text{Vs}$ )	1000	1500	2200	900
Hole mobility ( $\text{cm}^2/\text{Vs}$ )	40 to 100	100 to 500	1600	150
Young's modulus (GPa)	300 to 500	130 to 180	1000	200 to 300
Melting point ( $^{\circ}\text{C}$ )	2830*	1410	1400**	2400
Thermal conductivity ( $\text{Wcm}^{-1}\text{K}^{-1}$ )	5	1.5	20	1.3
Chemical Inertness	Excellent	Poor	Good but burn	Good
MEMS compatibility	Good	Excellent	Poor	Fair
Availability/Cost	Fair to good	Excellent	Poor	Fair

# Piezoresistive Sensors in Common Semiconductors



**Stable metal contacts and packaging become more challenging**

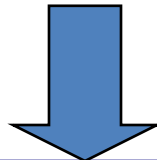
Hoang-Phuong Phan et al., "The Piezoresistive Effect of SiC for MEMS Sensors at High Temperatures: A Review," JOURNAL OF MICROELECTROMECHANICAL SYSTEMS, VOL. 24, NO. 6, DECEMBER 2015.





## Technological Challenges

- Thermomechanical induced stress in package
- Reaction kinetics at contact metallization to device



### Failure mechanisms:

- Cracking due to CTE mismatch;
- Die attach delamination;
- Wire bond weakening/detaching due to:
  - inter-metallic diffusion;
  - low temperature eutectic formation;
  - vibration-driven fatigue;
- Unstable electrical contact resistance.



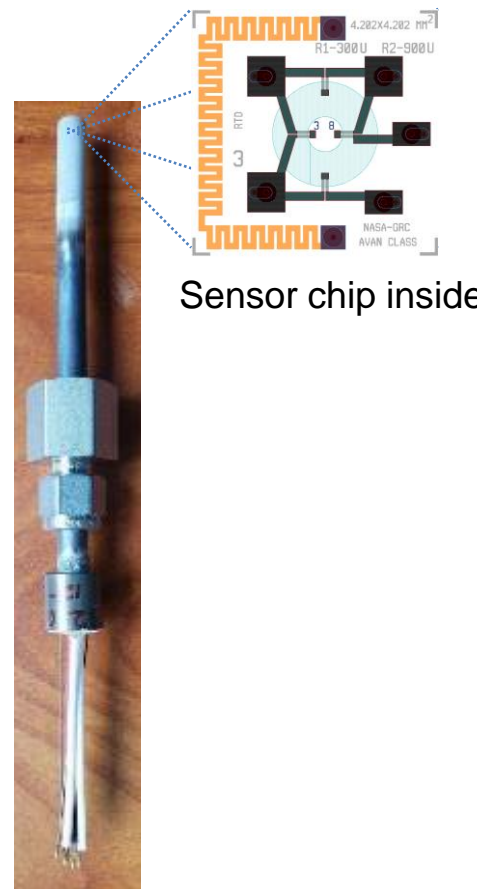
Electrical and Mechanical Failure

# Integrated Pressure/Temperature Sensor for 800 °C Operation

**Integrated Pressure/Temp Sensors at 800 °C without Cooling**

**Accurate Pressure/Temp Relationship, Real-time Temperature Compensation and Voltage-Pressure Conversion.**

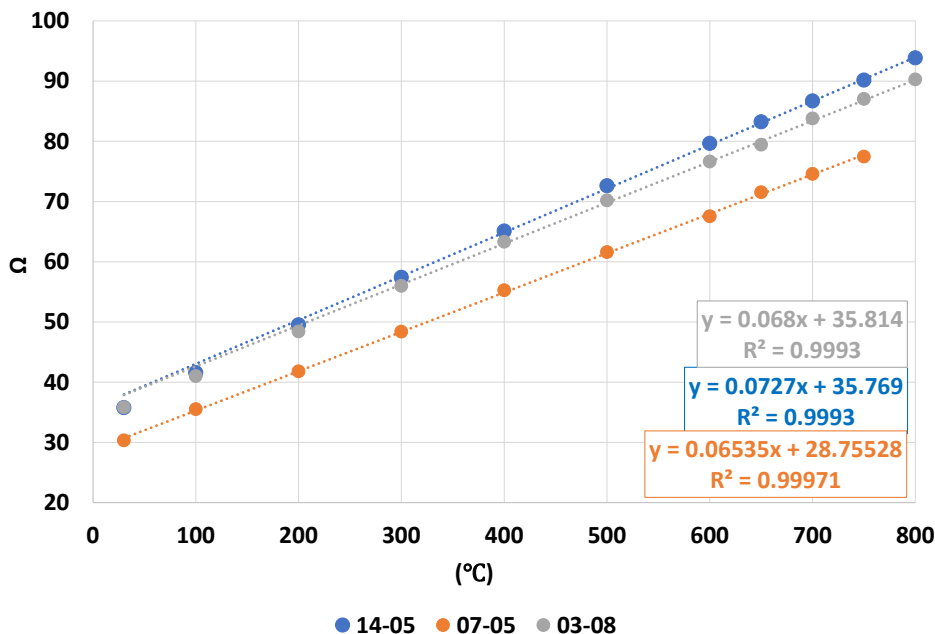
**Full-bandwidth Capture of Pressure Transient due to Direct Interaction with Flow-Field at High Temperature.**



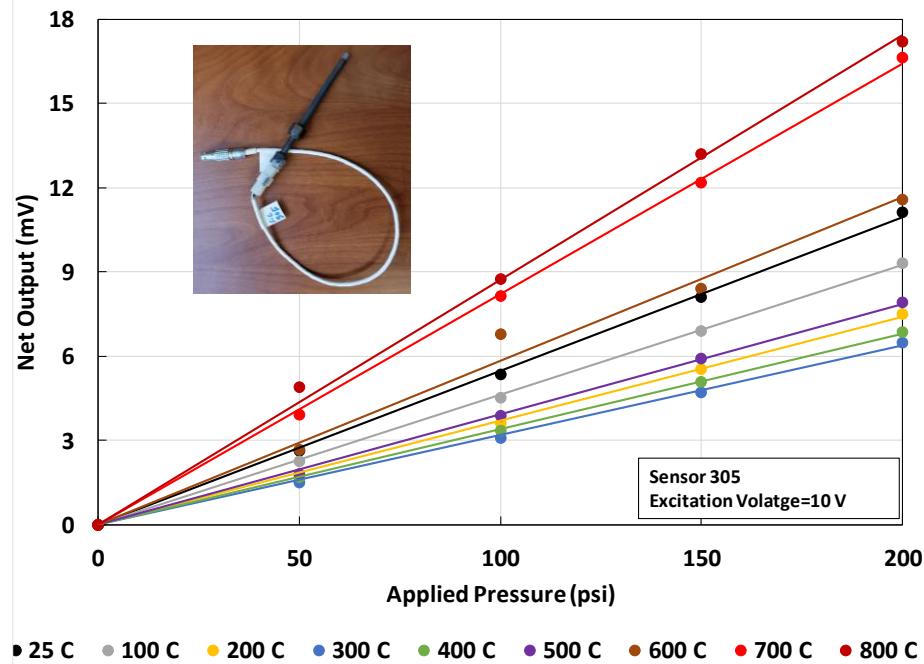
**Unique Characteristic: No wire bond Direct Chip Attach**

# SiC Pressure Sensor Operation at 800 °C

HG0767-04 RTD versus Temperature after 1 hr Burn-In at 800 °C



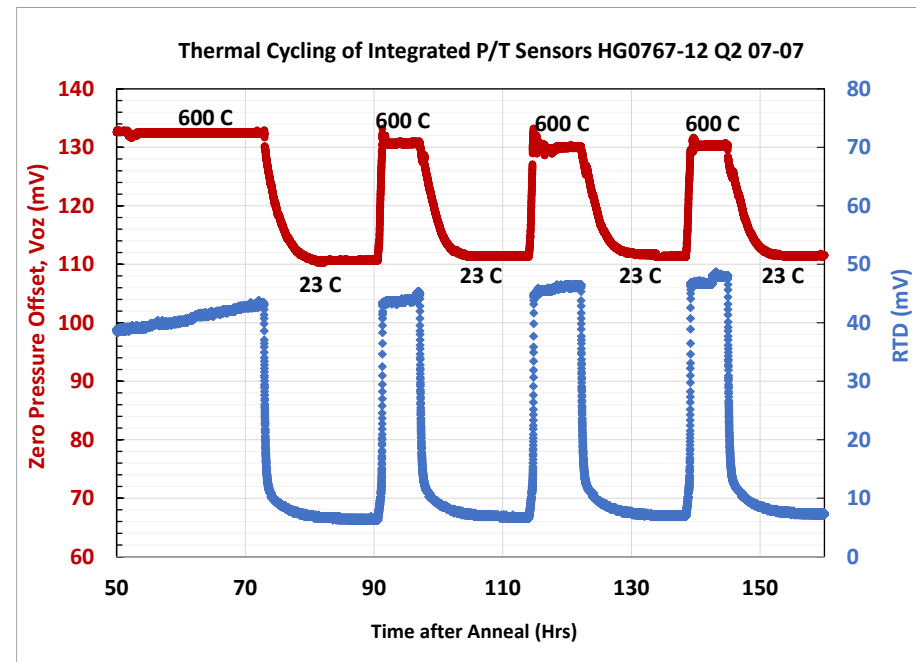
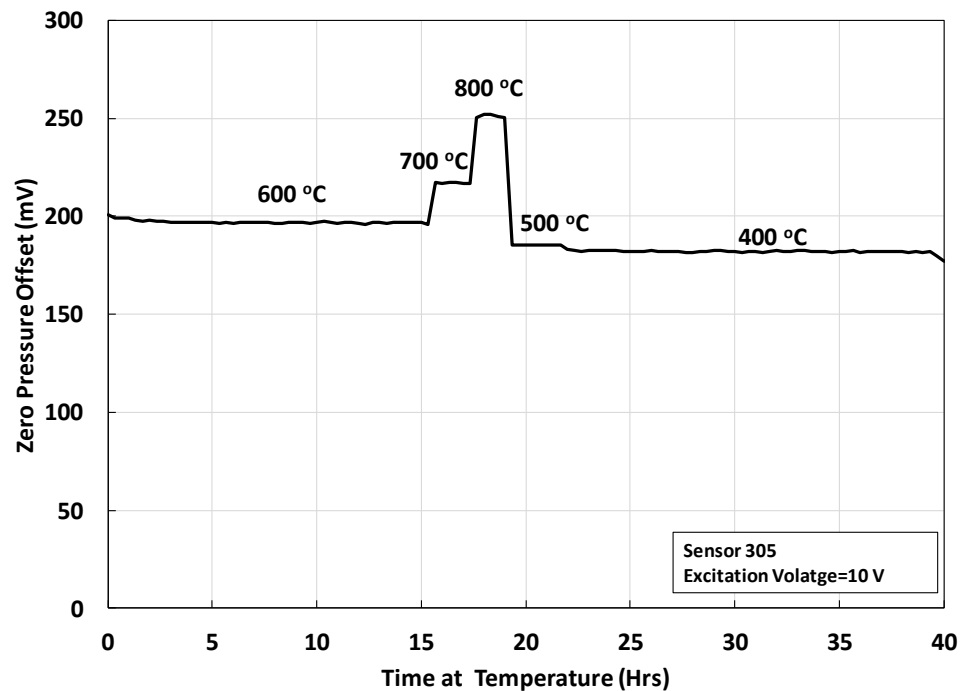
Temperature Sensors



Pressure Sensor



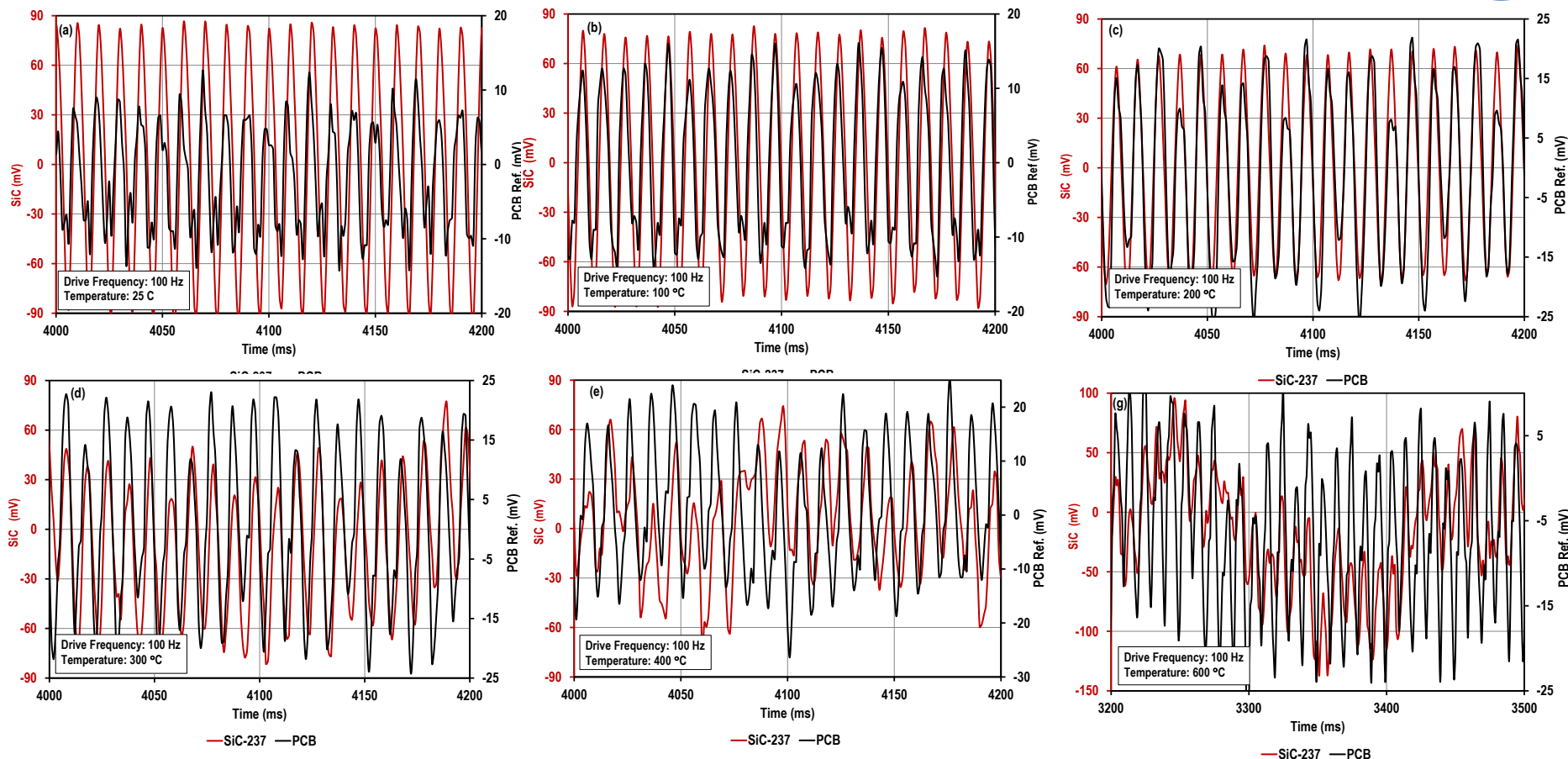
# Thermal Cycling Characteristics



Okojie et al., The International Conference on Solid-State Sensors, Actuators and Microsystems Transducers 2019 - EUROSENSORS XXXIII  
Berlin, GERMANY, 23-27 June 2019

- *Rise-time spikes due to oven over temperature excursion*
- *Temperature sensor requires further burn-in to stabilize*

# Dynamic Characterization-Time Domain

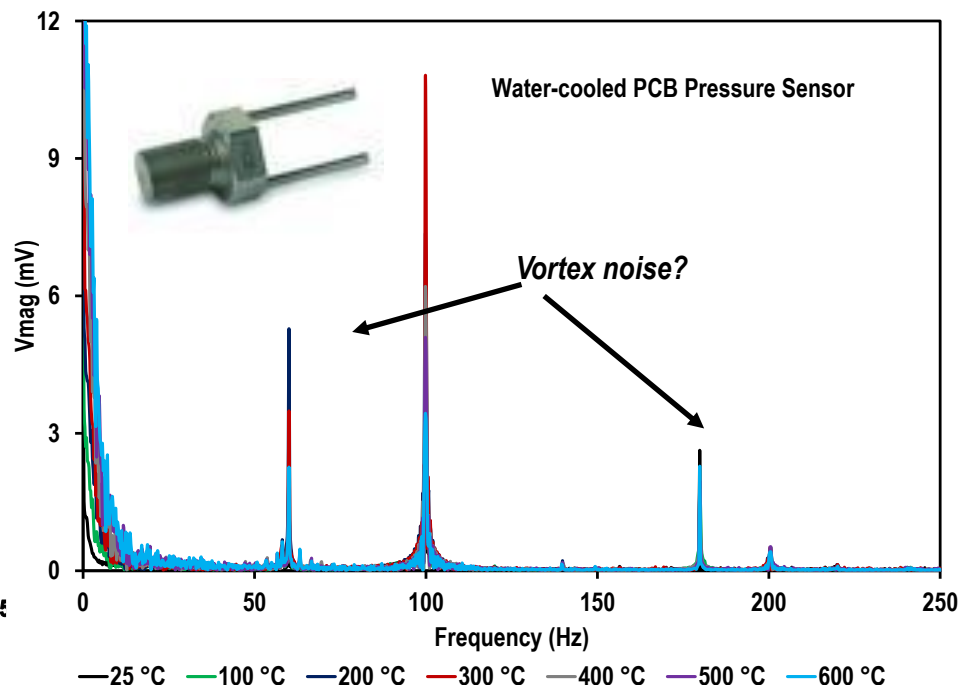
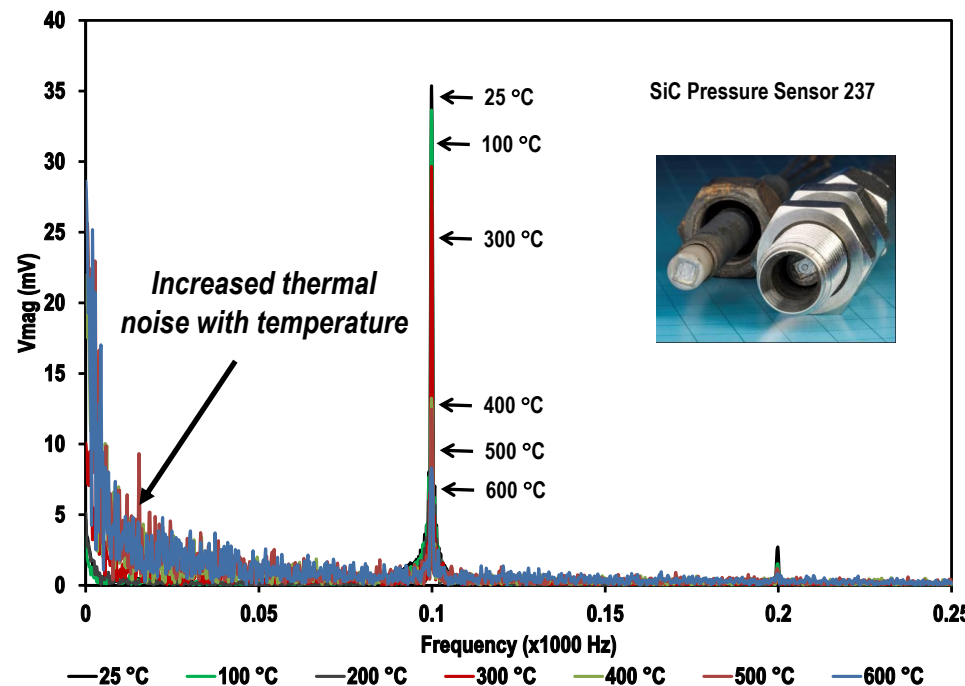


**Brief time history of the static SiC and dynamic piezoceramic pressure sensors in response to 100 Hz pressure modulation. Gain=100; Excitation voltage=10 V.**

$$\text{Equivalent } \Delta p = (\text{Dynamic Output} / \text{Gain}) / [\text{Pressure Sensitivity (T)} * \text{Input Voltage}]$$

$$\Delta p (\text{Avg}) \sim 1.88 \text{ psi}$$

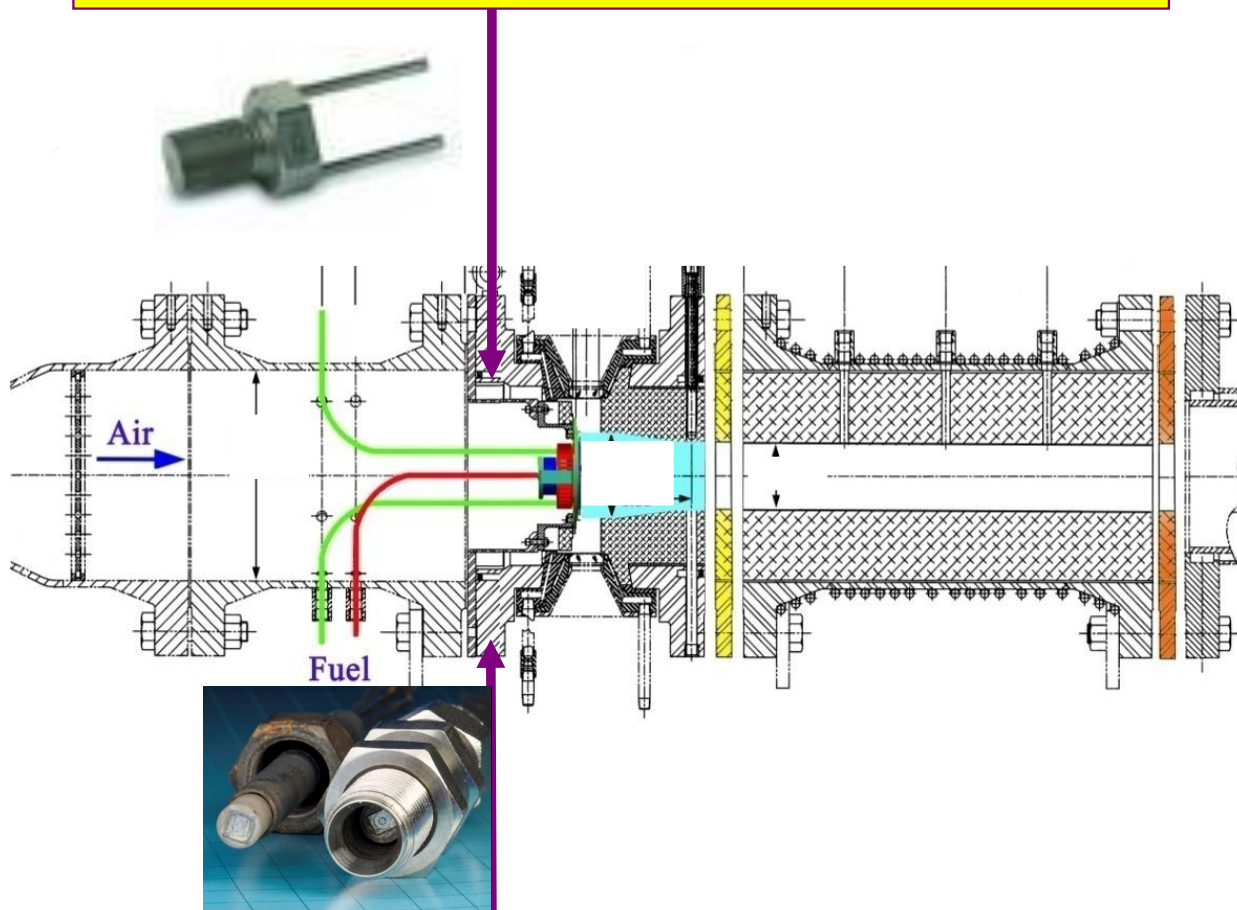
# Dynamic Characterization-Frequency Domain



**Amplitude magnitude response of the static SiC pressure sensor and the water-cooled dynamic piezoceramic pressure sensor at 100 Hz pressure modulation.**

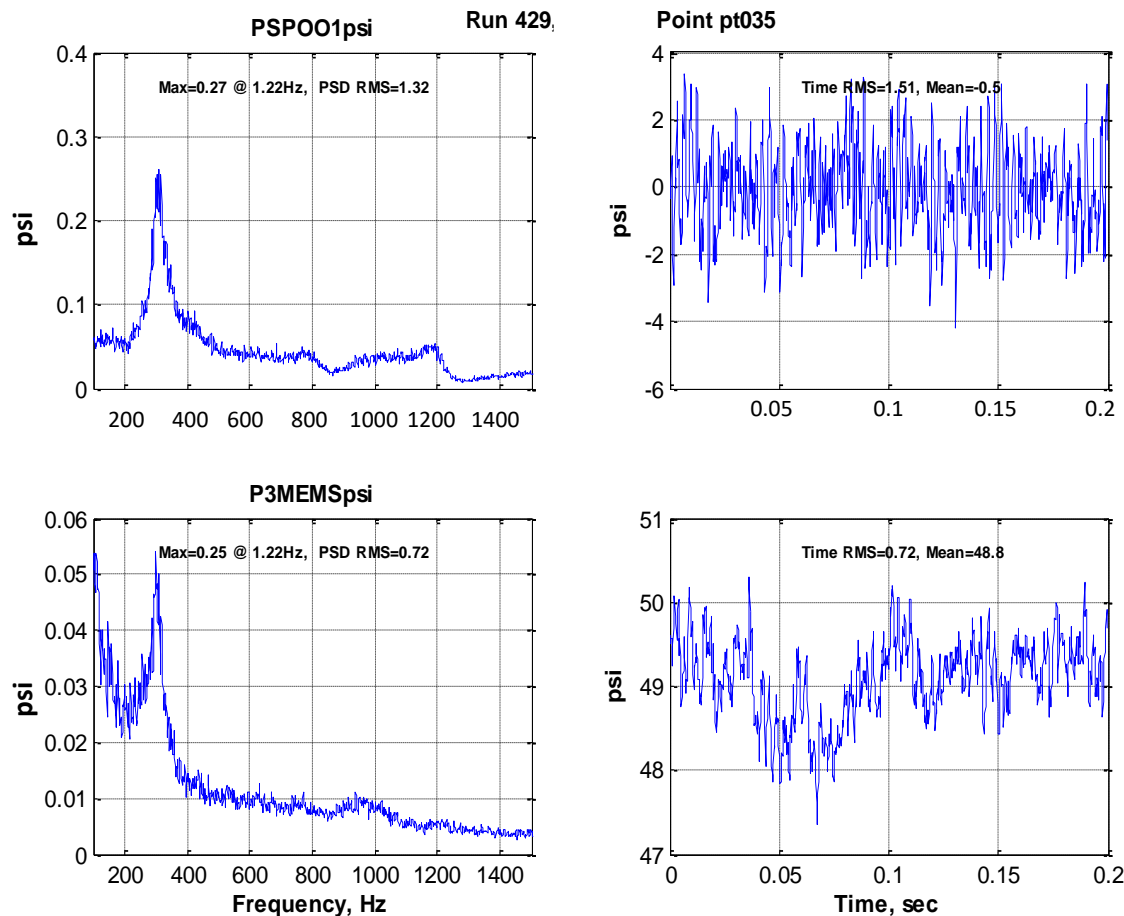
## Field Validation: Combustor Rig Test

PCB dynamic pressure transducers mounted on semi-infinite line



Flush mounted SiC dynamic pressure transducer

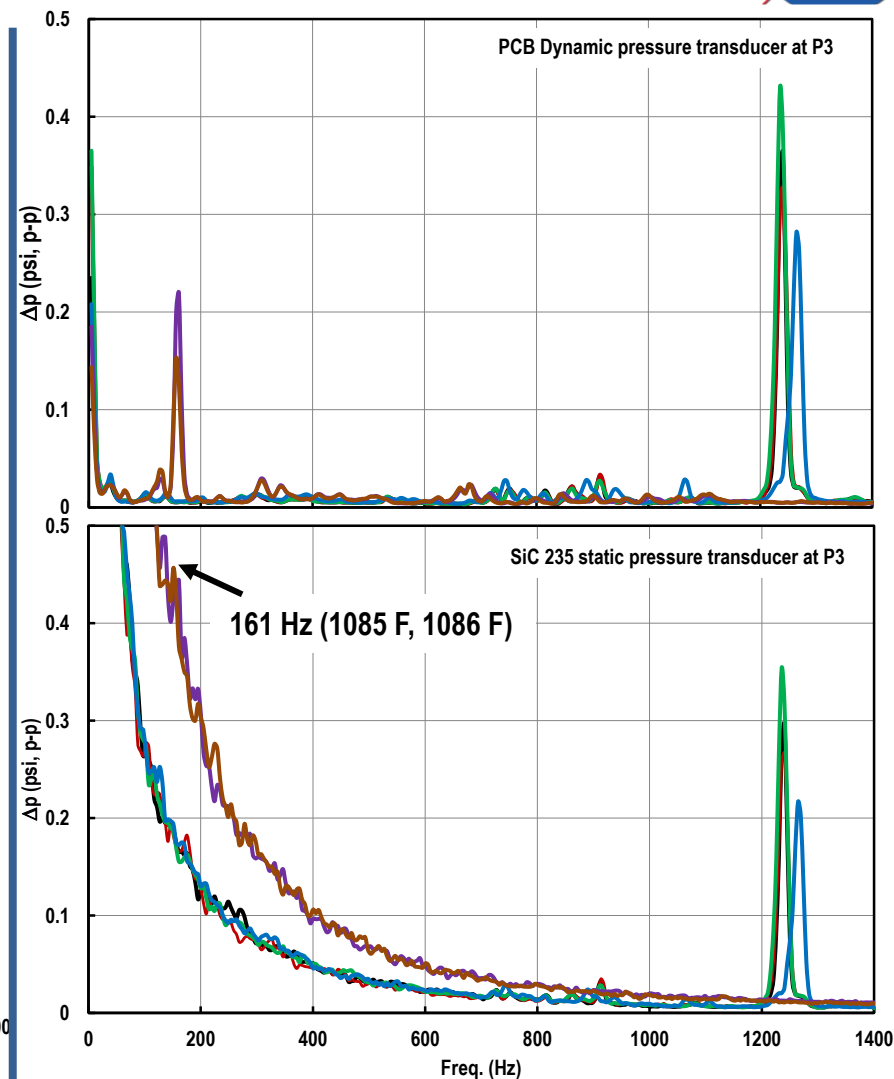
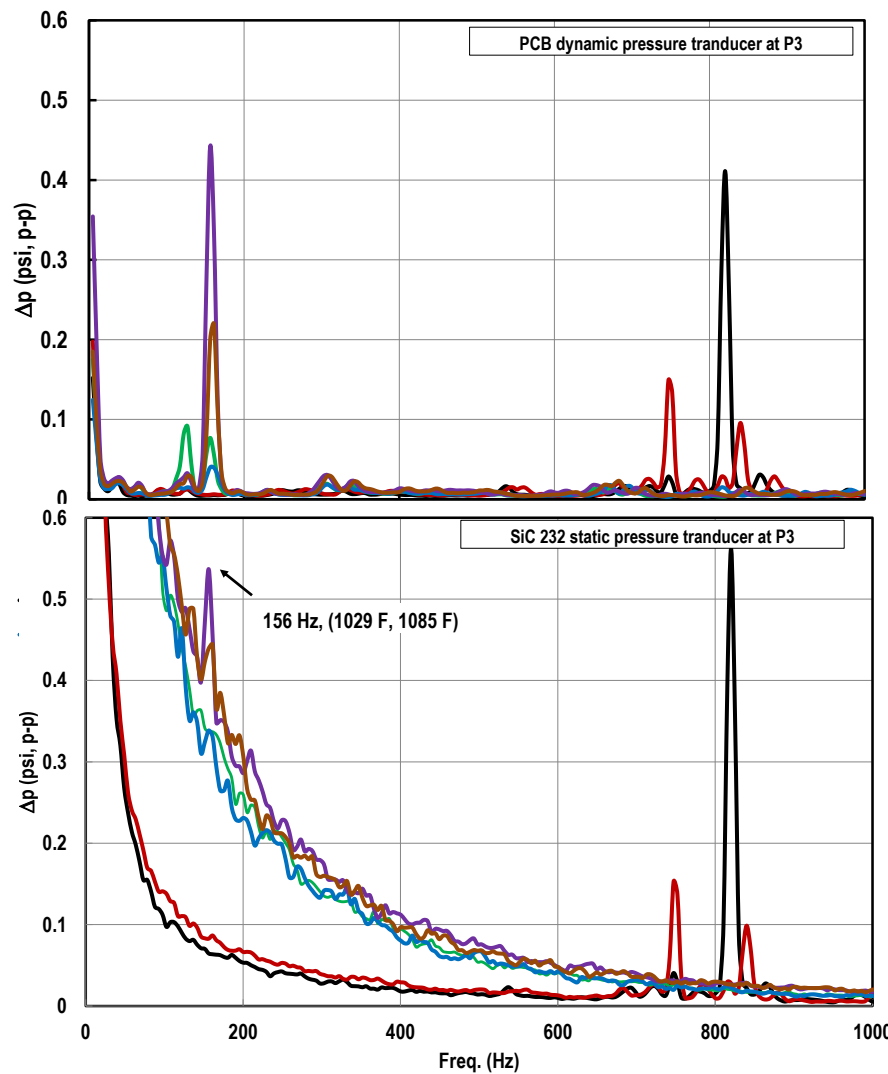
# Field Validation: Combustor Rig Test-1



**Amplitude magnitude (left column) and brief time history (right column) for the piezoelectric pressure transducer and the SiC pressure transducer shows the detection of thermo-acoustic instability at 310 Hz by both devices.**

The 13th International Conference on Solid-State Sensors, Actuators and Microsystems, Seoul, Korea, June 5-9, 2005

# Field Validation: Combustor Rig Tests 2 and 3

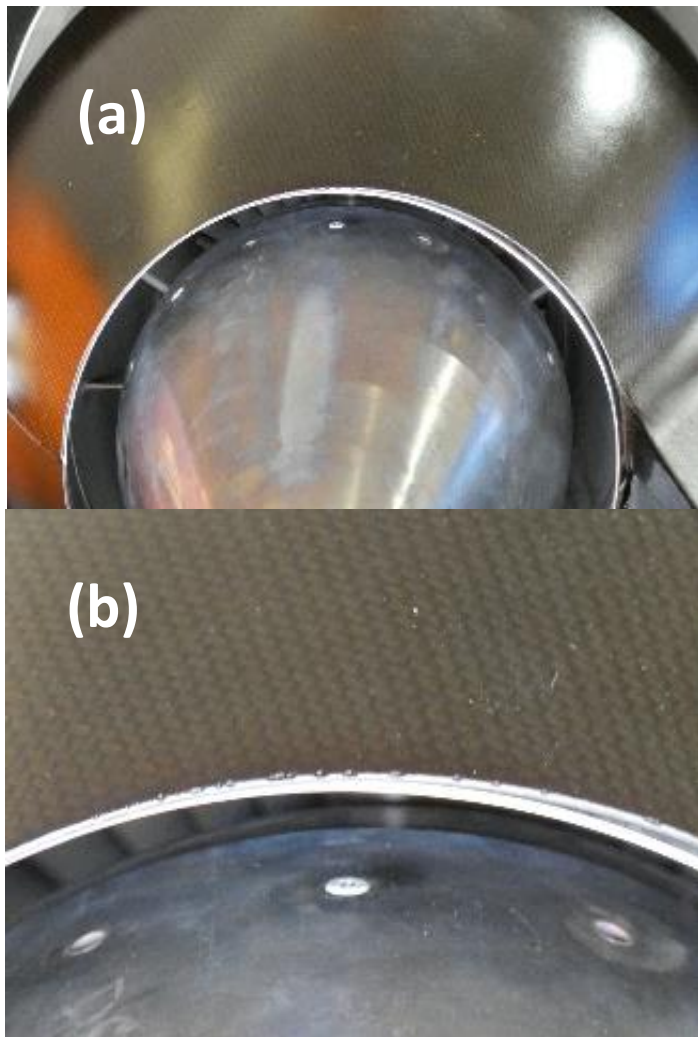


— 24.06.2013 14-12-15 (502 F) — 24.06.2013 15-23-53 (508 F) — 25.06.2013 14-24-34 (1026 F)  
 — 25.06.2013 14-14-26 (1027 F) — 25.06.2013 16-10-53 (1029 F) — 26.06.2013 18-33-07 (1085 F)

— 26.06.2013 15-11-19 (703 F) — 26.06.2013 15-11-50 (703 F) — 26.06.2013 15-12-29 (703 F)  
 — 26.06.2013 15-46-49 (703 F) — 26.06.2013 18-33-07 (1085 F) — 26.06.2013 18-34-11 (1086 F)

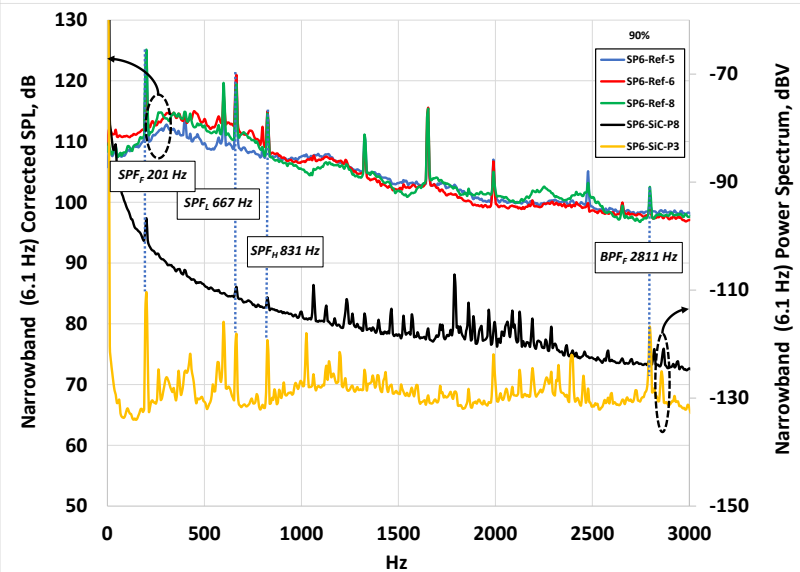
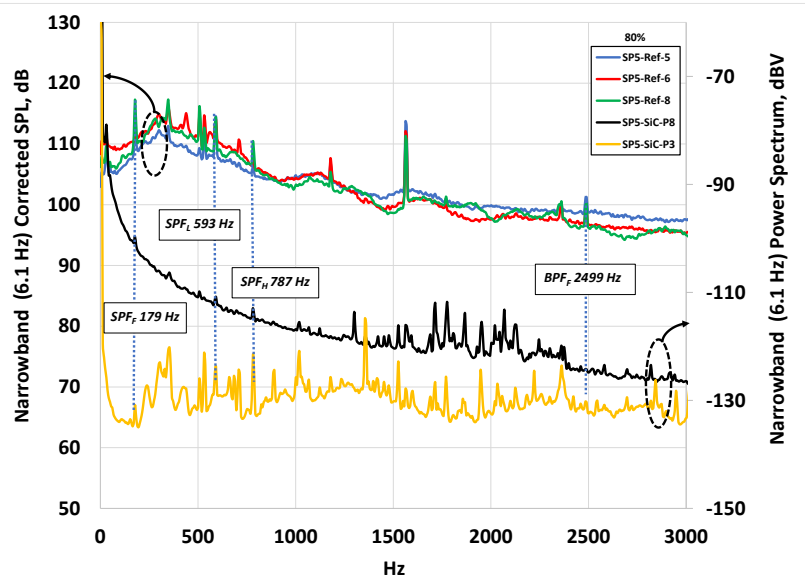
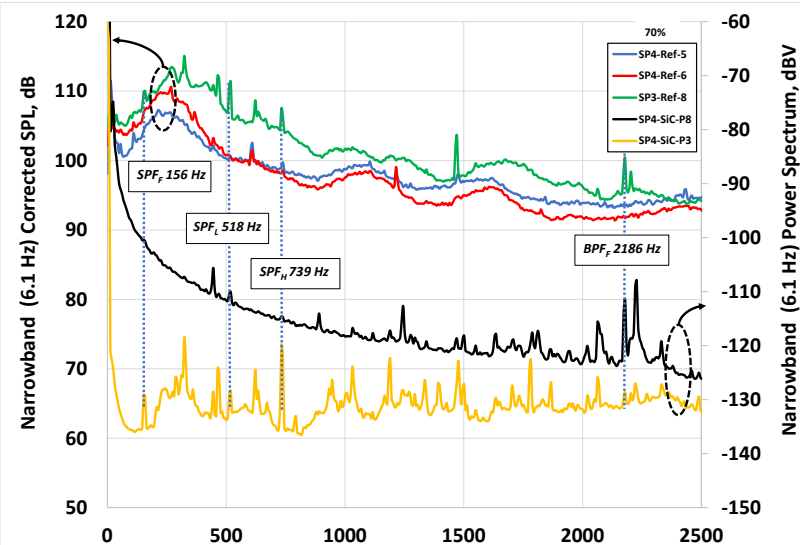
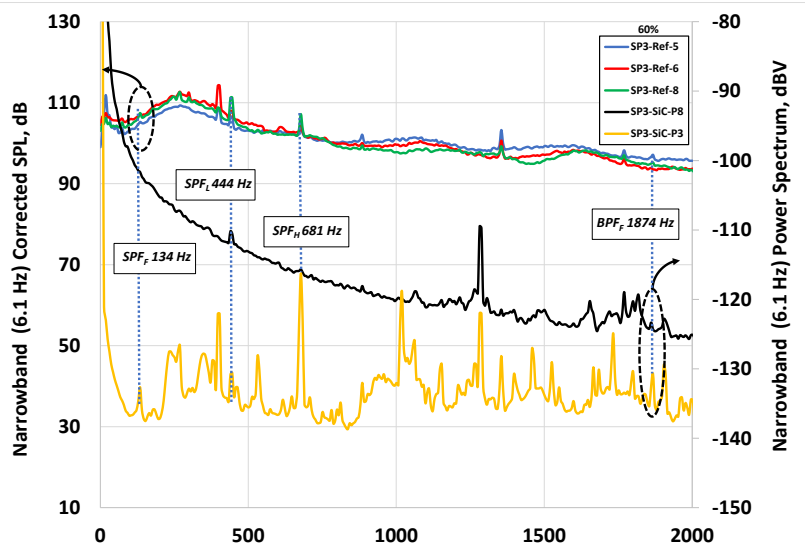


# Jet Engine Test: Flush-Mounted Versus Infinite Tube Probe



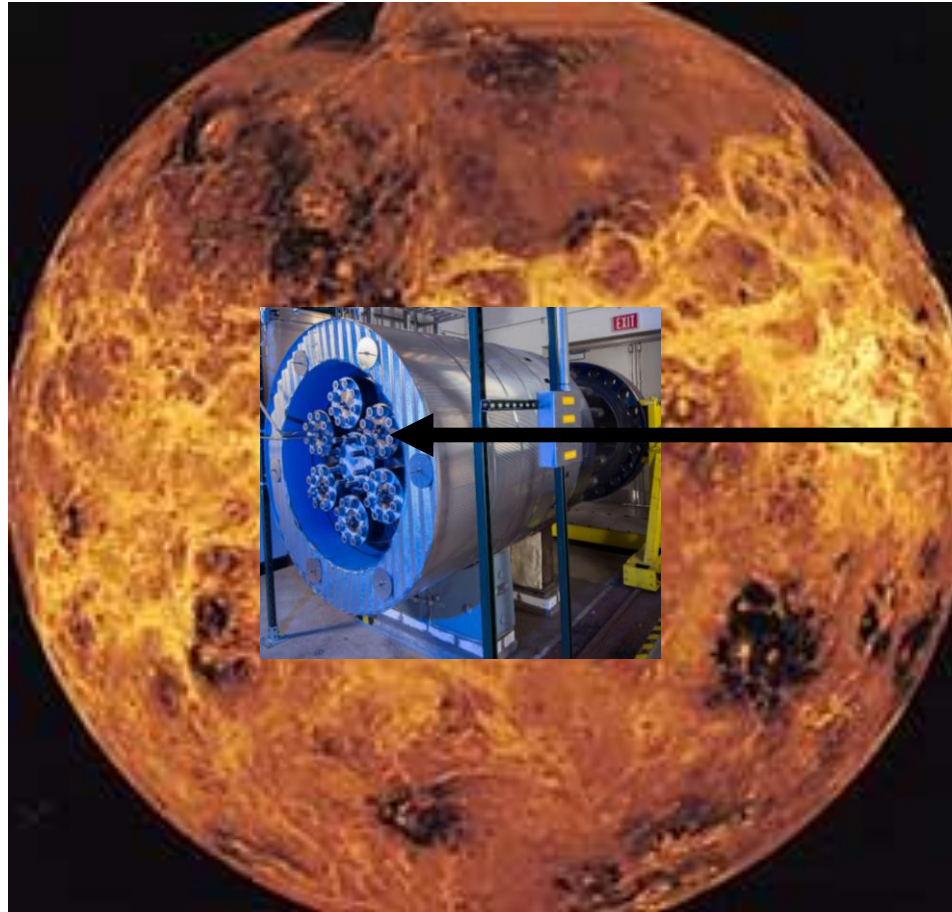
**Core exhaust nozzle of NASA DGEN experimental engine showing circumferentially placed pressure ports with open ITP hole terminations, b) flush-mounted SiC pressure sensor between two open ITP holes, and c) benchmark pressure sensors attached to ITPs external to the engine.**

# Flush-Mounted Versus Infinite Tube Probe (*Engine Test Results*)



**Good agreement between flush-mounted SiC pressure sensors and ITP-mounted commercial pressure sensor.**

# Test in Simulated Venus Environment



NASA GRC Extreme  
Environment Rig  
(GEER)

Temperature: 460 C

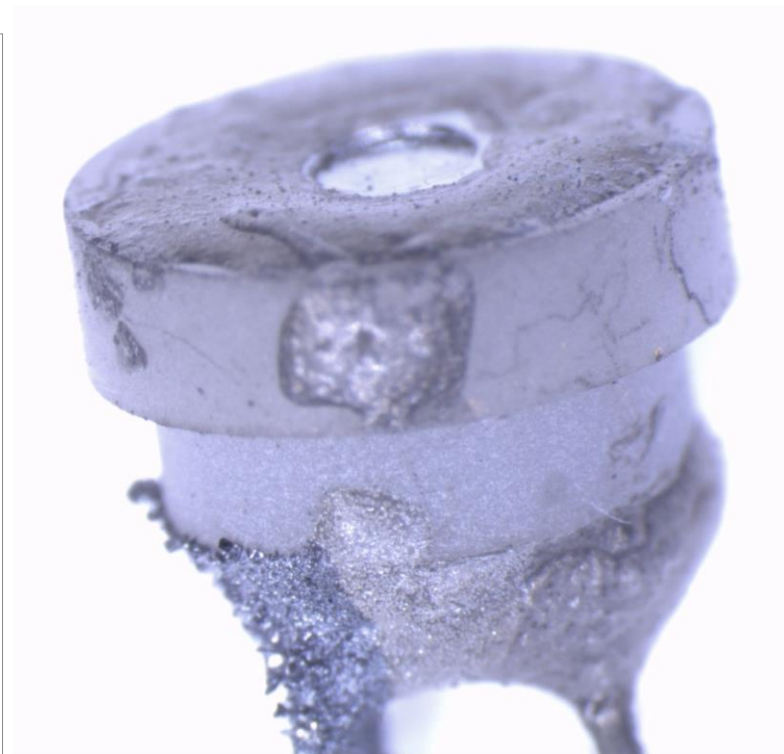
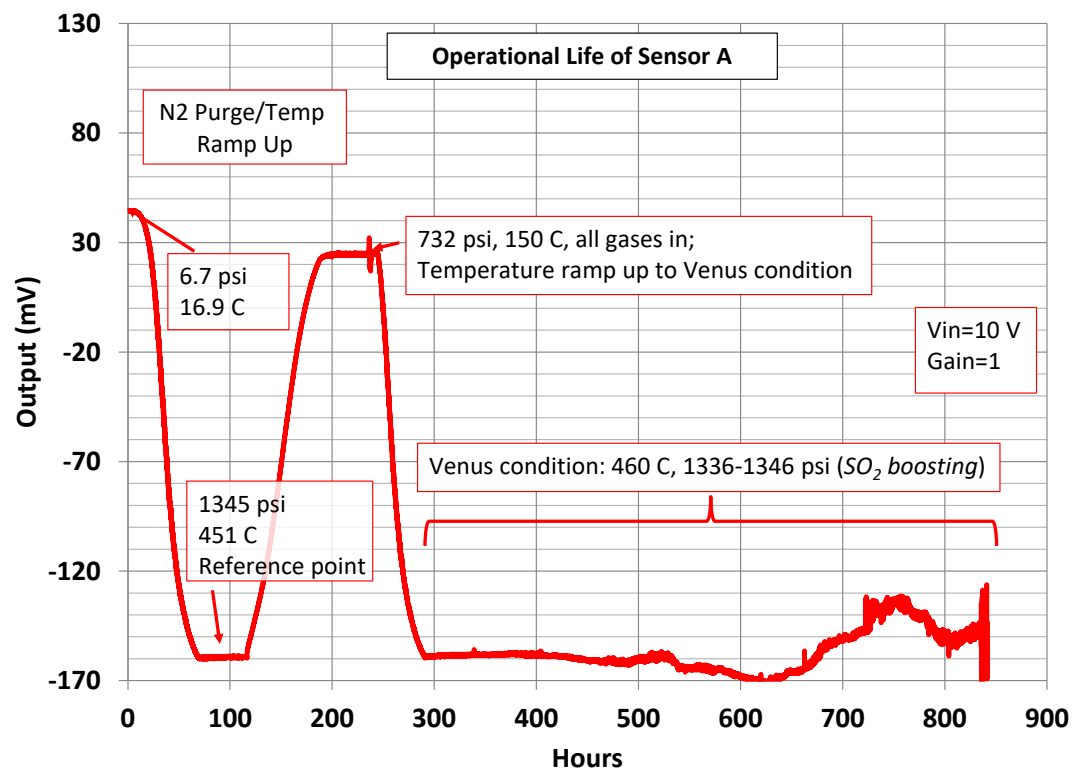
Pressure: 1356 psia

Chemistry: "Venus atmosphere" 96.5% CO<sub>2</sub>, 3.5% N<sub>2</sub>, 180 ppm

SO<sub>2</sub>, 12 ppm CO, 51 ppm OCS, 2 ppm H<sub>2</sub>S, 0.5 ppm HCl, 2.5 ppb

HF Range: SO<sub>2</sub> concentration range 100-200 ppm

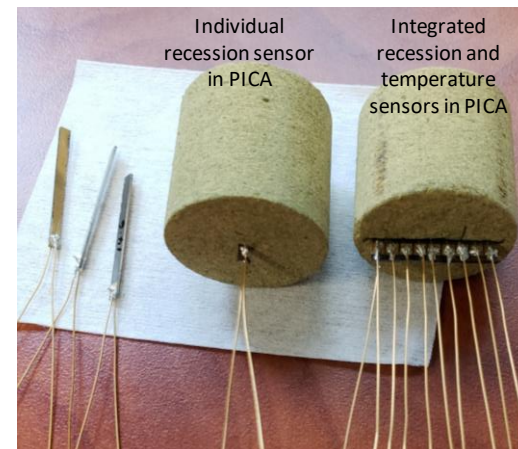
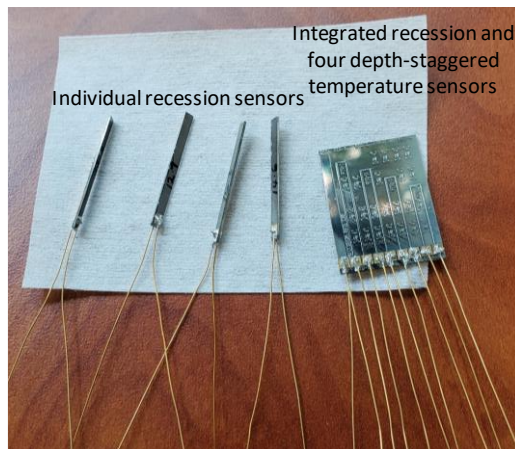
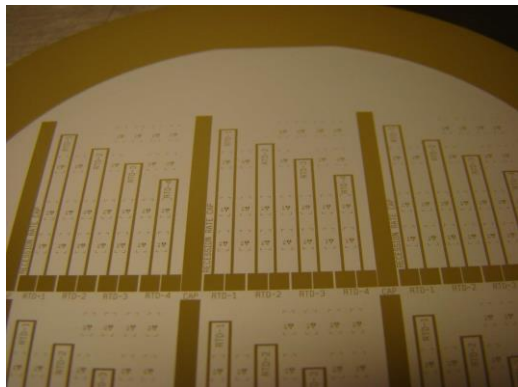
# SiC Pressure Sensor Venus Life in GEER



- Sensor reliably operational for ~12 earth days
- Sensor survived 60 earth days
- No mechanical damage
- Reaction chemistry on exposed wires caused failure



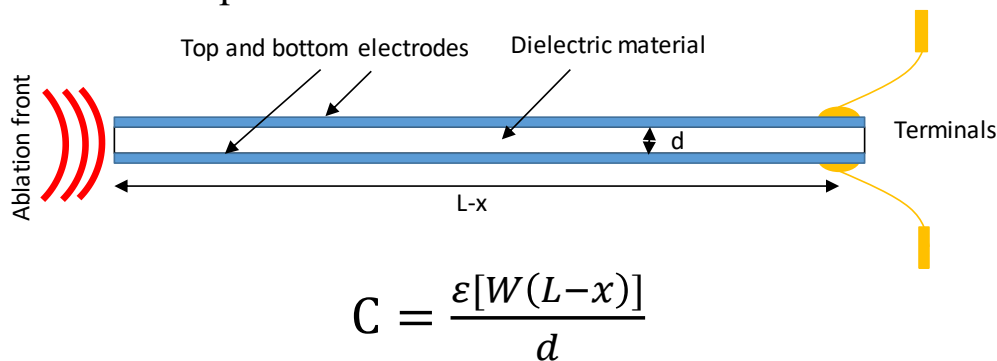
# Minimally Intrusive Integrated Recession and Temperature Sensors



- Batch fabrication provides uniform measurement
- Speeds up production
- Reduces costs and reduces human error during sensor placement.

- Minimally intrusive insertion into the TPS heatshield.
- Large area distribution of sensors leads to improved health monitoring due to high fidelity.
- Eliminates need for plugs and inserted directly into the heatshield fore body at point of TPS manufacture.

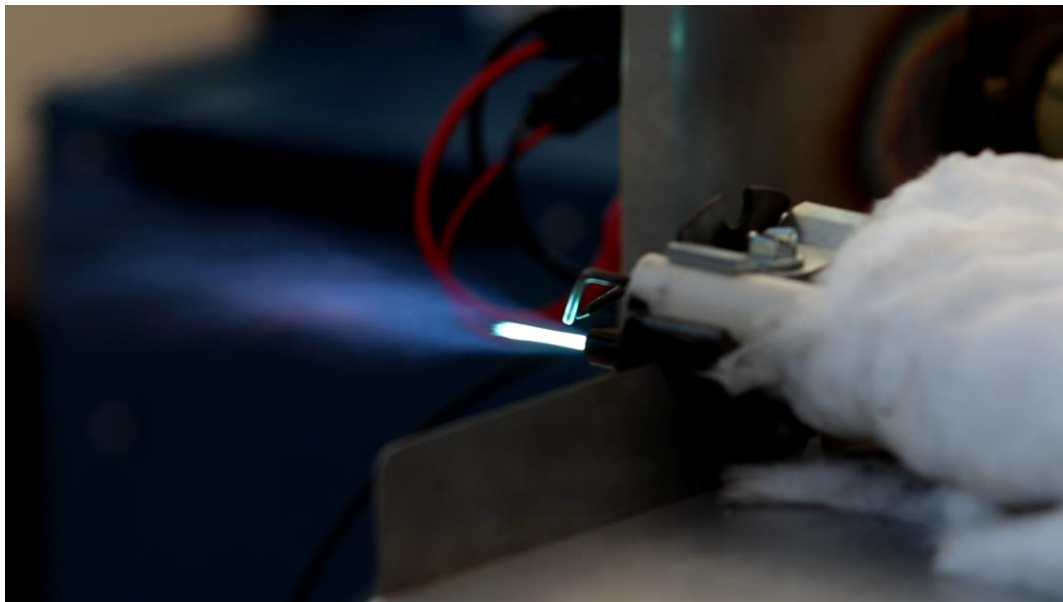
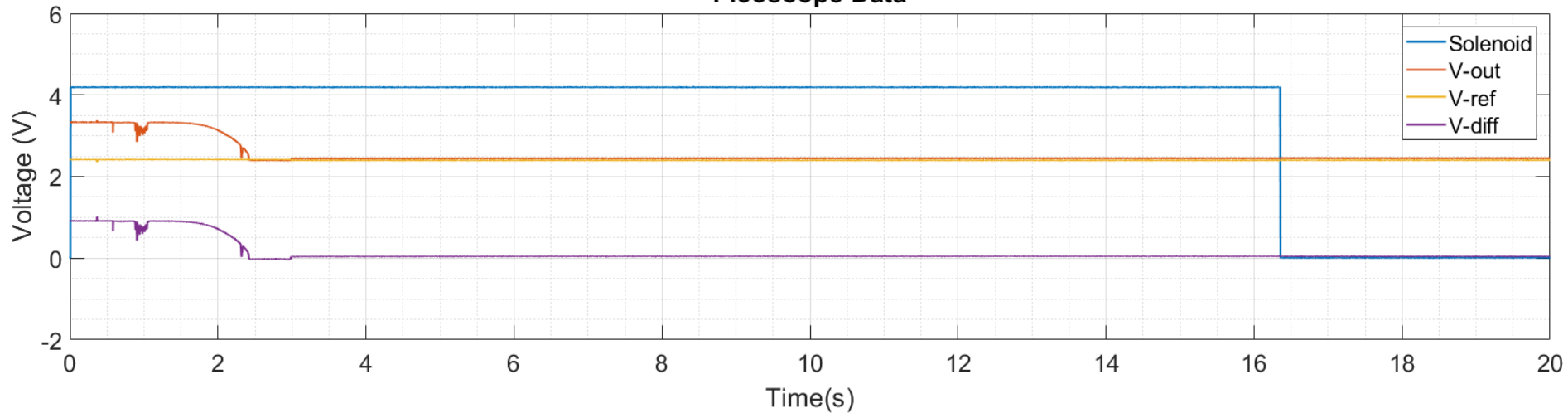
Capacitive recession sensor



Provides real-time monitoring recession due to ablation, attrition, wear, etc.

# Recession Sensor Test Results

## D02 Test Data Summary Picoscope Data



**Voltage output converted to corresponding material loss**





## Conclusion

- ✓ Successfully developed SiC piezoresistive pressure sensors capable of operating up to 600 °C for hundreds of hours, and short duration
- ✓ The successful development was the result of years of efforts to overcome the problems of
  - increasing voltage offset drifts with increasing temperature that renders sensors useless.
  - Thermo-mechanically induced failures due to the earlier application of conventional packaging
- ✓ Thermally stable ohmic contacts to SiC resulted in solving the drift problem
- ✓ Introduction of MEMS DCA packaging solved the thermo-mechanical failures
- ✓ Because of this work, it is **now possible** to **apply the SiC pressure sensors** for **longer duration** engine ground tests and potential **short duration flight** tests in high temperature regimes.
- ✓ First generation capacitive recession sensors demonstrated.
- ✓ Applications include real time monitoring of ablation, abrasion, attrition, and wear.

